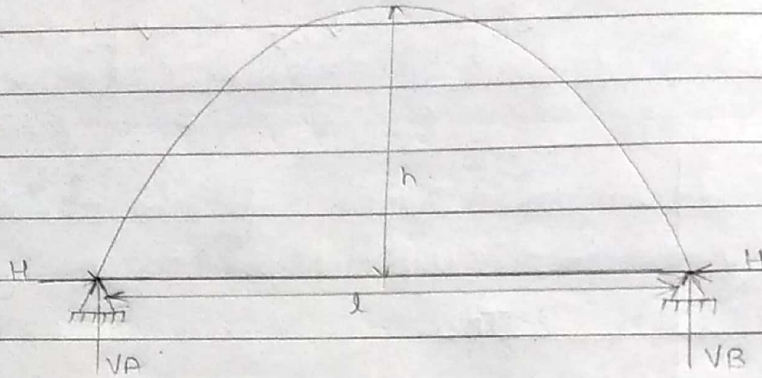


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Unit 4 Analysis of Column and Arches

[2-3 hinge]

* Analysis of 2 hinge arches



where H = Horizontal thrust

$$H = \int_0^l \frac{My dx}{y^2 dx}$$

where $y = \frac{4hx}{l^2} (l-x)$

$$= \int_0^l \frac{16h^2 x^2 (l-x)^2 dx}{l^4}$$

$$= \int_0^l \frac{16h^2 x^2 (l^2 - 2lx + x^2) dx}{l^4}$$

$$= \frac{16h^2}{l^4} \int_0^l x^2 (l^2 - 2lx + x^2) dx$$

$$= \frac{16h^2}{l^4} \int_0^l (x^2 l^2 - 2lx^3 + x^4) dx$$

$$= \frac{16h^2}{l^4} \left[l^2 \int_0^l x^2 dx - 2l \int_0^l x^3 dx + \int_0^l x^4 dx \right]$$

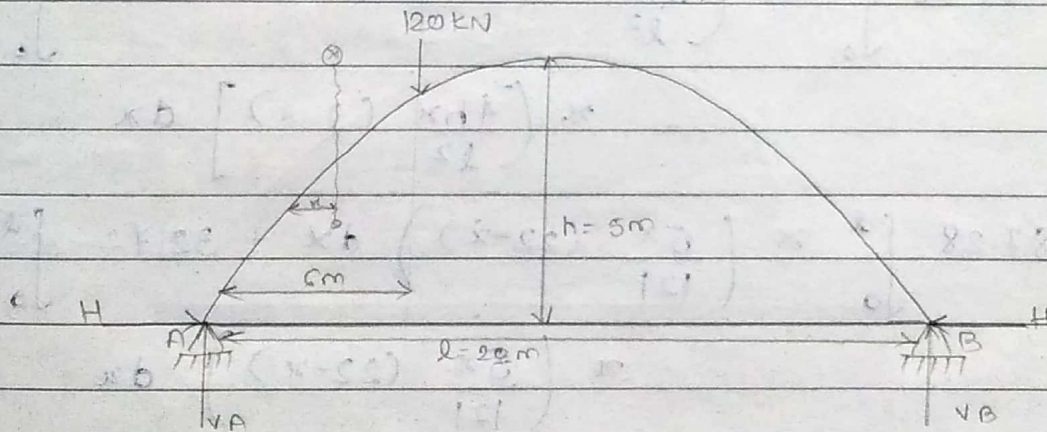
$$= \frac{16h^2}{l^4} \left[l^2 \left(\frac{x^3}{3} \right)_0^l - 2l \left(\frac{x^4}{4} \right)_0^l + \left(\frac{x^5}{5} \right)_0^l \right]$$

$$= \frac{16h^2}{l^4} \left[l^2 \cdot \frac{l^3}{3} - 2l \times \frac{l^4}{4} + \frac{l^5}{5} \right]$$

$$= \frac{16h^2}{l^4} \left[\frac{l^5}{3} - \frac{2l^5}{4} + \frac{l^5}{5} \right] = \frac{16h^2 l^5}{l^4} \left(\frac{1}{30} \right) = \frac{8}{15} h^2 l$$

* Numerical *

Q Two hinge parabolic arches of span 22m and the central rise 5m carries the point load of 120 kN, 6m from the left hand support. Analyzed the arches and determine the horizontal thrust.



$$\sum F_y = 0$$

$$V_A + V_B = 120\text{ kN}$$

$$\sum M @ A = 0$$

$$120 \times 6 - V_B \times 22 = 0$$

$$V_B = 32.73\text{ kN}$$

$$V_A = 87.28\text{ kN}$$

$$y = \frac{4hn}{l^2} (l-x)$$

$$= \frac{4 \times 5n}{22^2} (22-n)$$

$$= \frac{20n}{22^2} (22-n) = \frac{110n - 5n^2}{121} \text{ PAGE No: } \underline{\quad}$$

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Portion	Origin	Limit	M
AC	A	0-6	87.28x
BC	B	0-16	32.78x

$$H = \int_0^l \frac{My dx}{y^2 dx}$$

$$= \int_0^l My dx$$

$$= \int_0^l (87.28x) y dx + \int_0^l (32.78x) y dx$$

$$= 87.28 \int_0^l x \left[\frac{4hx}{l^2} (l-x) \right] dx + 32.72 \int_0^l$$

$$x \left[\frac{4hx}{l^2} (l-x) \right] dx$$

$$= 87.28 \int_0^l x \left[\frac{5x}{121} (22-x) \right] dx + 32.72 \int_0^l$$

$$x \left[\frac{5x}{121} (22-x) \right] dx$$

$$= 87.28 \int_0^6 x (0.90x - 0.04x^2) dx + 32.72 \int_0^{16}$$

$$x (0.90x - 0.04x^2) dx$$

$$= 87.28 \left[\int_0^6 0.90x^2 dx - \int_0^6 0.04x^3 dx \right] +$$

$$32.78 \left[\int_0^{16} 0.90x^2 dx - \int_0^{16} 0.04x^3 dx \right]$$

$$= \int_0^6 87.28x \left(\frac{10x - 5x^2}{6} \right) dx + \int_0^{16} 32.78x \left(\frac{10 - 5x^2}{4} \right) dx$$

$$= \int_0^6 79.34x^2 - 3.6x^3 dx + \int_0^{16} 24.75x^2 - 1.33x^3 dx$$

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$$= 87.28 \times 0.90 \left(\frac{x^3}{3} \right)_0^6 - 0.04 \left(\frac{x^4}{4} \right)_0^6 + 32.78 \times$$

$$0.90 \left(\frac{x^3}{3} \right)_0^{16} - 0.04 \left(\frac{x^4}{4} \right)_0^{16}$$

$$= 87.28 \times 0.90 \times \left(\frac{6^3}{3} \right) - 0.04 \left(\frac{6^4}{4} \right) + 32.78 \times 0.90 \left(\frac{16^3}{3} \right)$$

$$- 0.04 \left(\frac{16^4}{4} \right)$$

$$= 87.28 \times 0.90 \times 72 - 0.04 \times 324 + 32.78 \times 0.90 \times 1365.33$$

$$- 0.04 \times 16384$$

$$= 4524.39 + 18797.36$$

$$= 23321.95$$

$$\int y^2 dx = \frac{8}{15} h^2 \times l$$

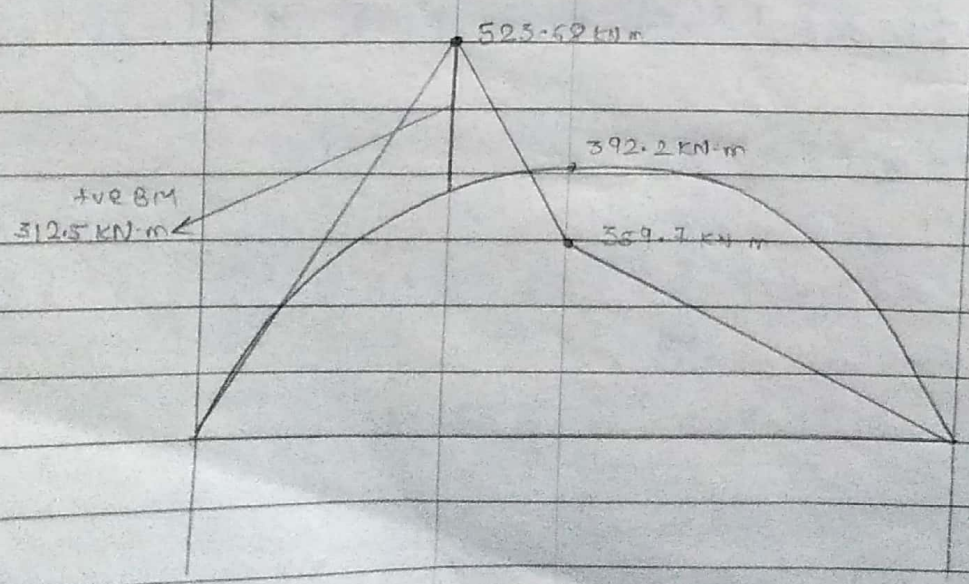
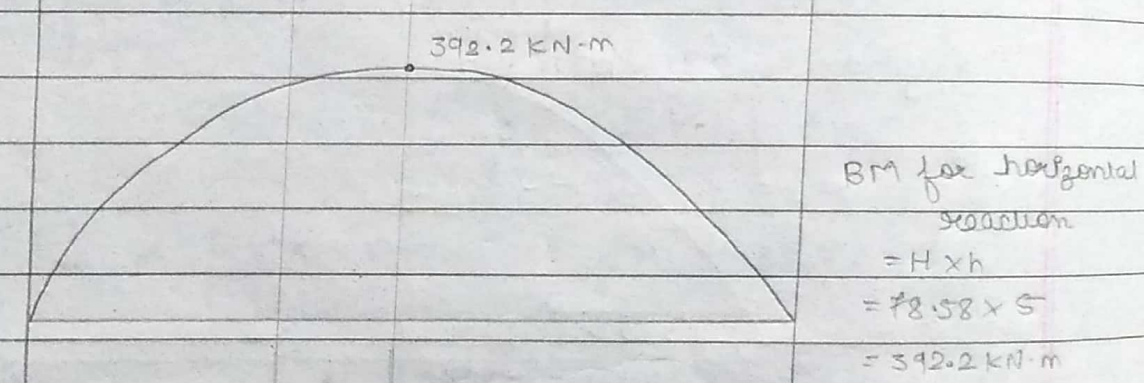
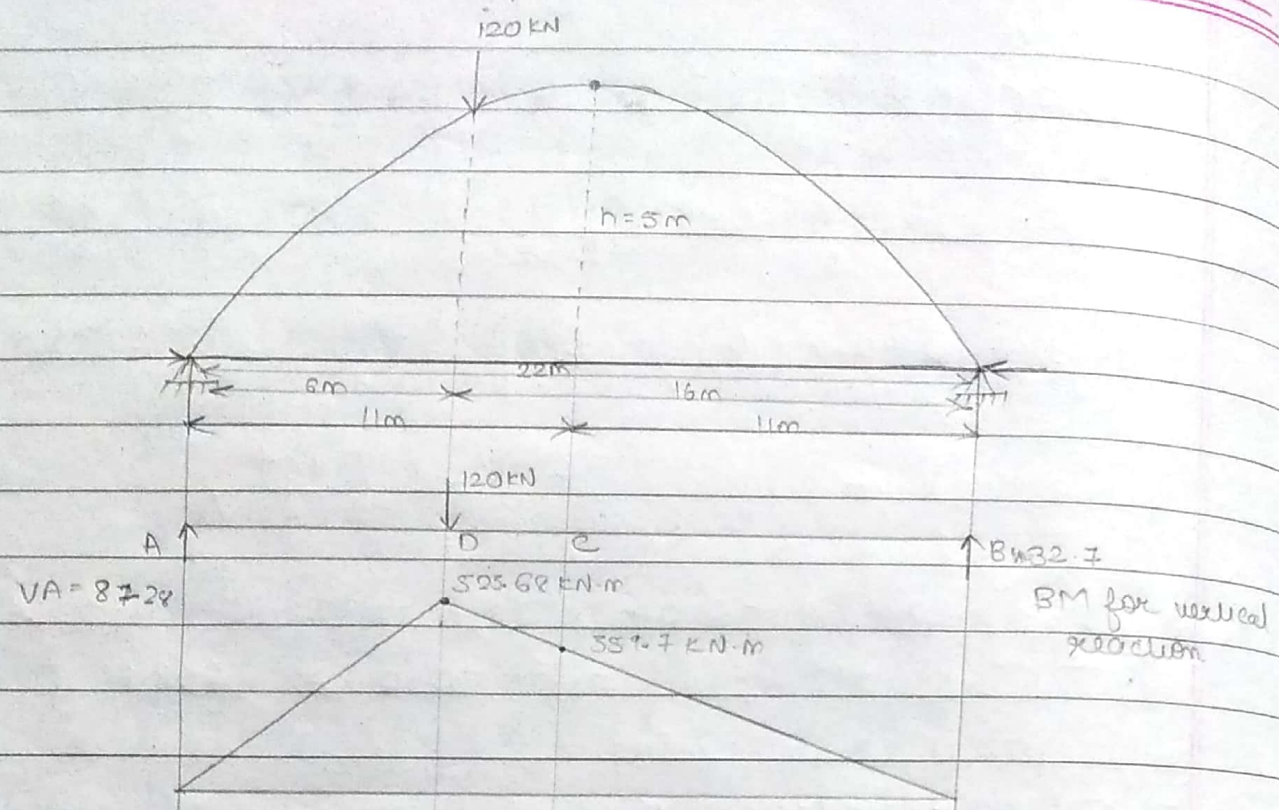
$$= \frac{8}{15} \times 5^2 \times 22$$

$$= 293.33$$

$$H = 23321.95$$

$$293.33$$

$$H = 79.50 \text{ kN}$$



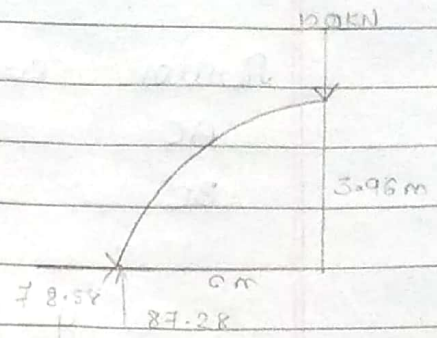
$$y = \frac{4hx}{l^2} (l-x)$$

$$= \frac{4h \times 6}{22^2} (22-6)$$

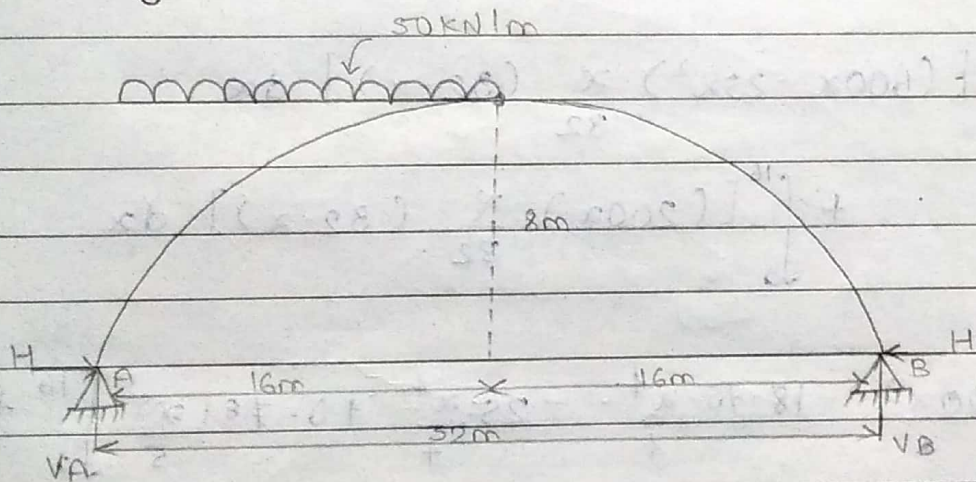
$$y = 3.96 \text{ m}$$

$$\text{Max}^m \text{ BM} = 87.28 \times 6 - 78.5 \times 3.96$$

$$\text{At A} = 212.5 \text{ KN-m}$$



Q A two hinge parabolic arch of span 22m with a central rise 8m carries the UDL of 50 kN/m over the left half span. Determine the horizontal thrust and plot the BMD profile.



$$\sum P_y = 0$$

$$V_A + V_B - 800 = 0$$

$$V_A + V_B = 800$$

$$\sum M \text{ at A} = 0$$

$$50 \times 16 \times \frac{16}{2} - V_B \times 32 = 0$$

$$32 V_B = 6400$$

$$V_B = 200 \text{ KN}$$

$$V_A = 600 \text{ KN}$$

Portion	Origin	Limit	Moment
AC	A	0-16	$600x - 50x^2$
BC	B	0-16	$200x$

$$y = \frac{hx}{l^2} (l-x)$$

$$= \frac{4 \times 8x}{32^2} (32-x)$$

$$y = \frac{1}{32} x(32-x)$$

$$= \int_0^l My \, dx$$

$$= \int_0^{16} \left[(600x - 25x^2) \frac{x}{32} (32-x) \right] dx$$

$$+ \int_0^{16} \left[(200x) \frac{x}{32} (32-x) \right] dx$$

$$= \left[\frac{600x^3}{3} - \frac{18.75x^4}{4} - \frac{25x^4}{4} + \frac{10.781x^5}{5} \right]_0^{16} +$$

$$\left[\frac{200x^3}{3} - \frac{6.25x^4}{4} \right]_0^{16}$$

$$= 819200 - 307200 - 409600 + 163787.57 + 273066.67 -$$

$$= 436854.24$$

$$\int y^2 \, dx = \frac{8}{15} h^2 l$$

$$= \frac{8}{15} \times 8^2 \times 32$$

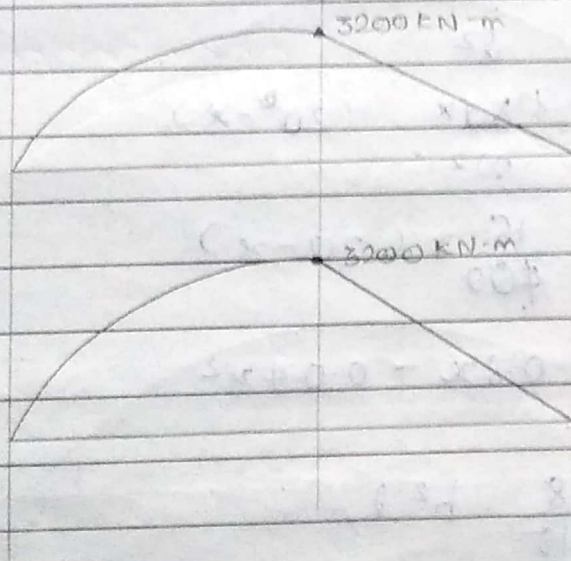
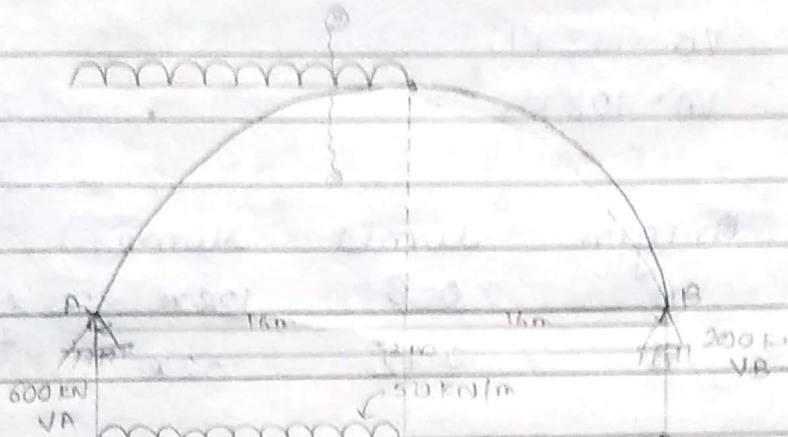
$$= 1092.26$$

$$H = \int \frac{M_y dx}{y^2}$$

$$= 486854.24$$

$$1092.26$$

$$H = 399.95 \text{ KN}$$

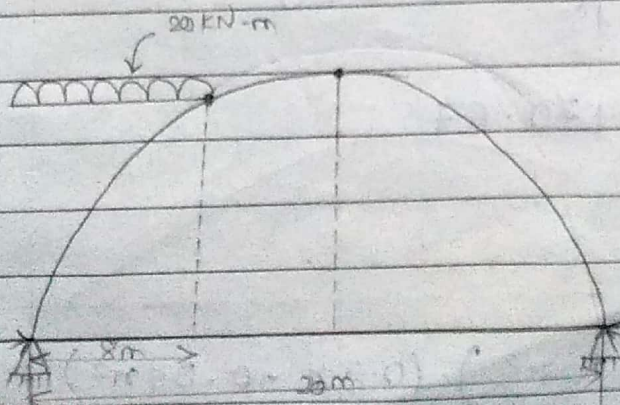


$$= 600 \times 16 - 50 \times 16 \times 8$$

$$= 3200 \text{ kN-m}$$

BM for vertical reaction

BM for horizontal reaction



$$H = 86.86 \text{ KN}$$

$$\Sigma F_y = 0$$

$$V_A + V_B - 20 \times 8 = 0$$

$$V_A + V_B = 160$$

$$\Sigma M @ A = 0$$

$$20 \times 8 \times \frac{8}{2} - V_B \times 20 = 0$$

$$V_B = 32 \text{ kN}$$

$$V_A = 128 \text{ kN}$$

Section	Origin	Limit	Moment
AC	A	0-8	$128x - \frac{20x^2}{2}$
BC	B	0-12	$32x$

$$y = \frac{4hx}{l^2} (l-x)$$

$$= \frac{4 \times 4x}{20^2} (20-x)$$

$$y = \frac{16}{400} x (20-x)$$

$$= 0.8x - 0.04x^2$$

$$y^2 = \frac{8}{15} b^2 l$$

$$= \frac{8}{15} \times 4^2 \times 20$$

$$= 170.67$$

$$\int My \, dx$$

$$\int_0^8 \left(\frac{128x - 20x^2}{2} \right) (0.8x - 0.04x^2) \, dx + \int_0^{12} 32x (0.8x - 0.04x^2) \, dx$$

$$\int_0^8 (128 - 10x^2)(0.8x - 0.04x^2) + \int_0^{12} 25.6x^2 - 1.28x^3$$

$$\int_0^8 102.4x^2 - 5.12x^3 - 8x^3 + 0.4x^4 + \int_0^{12} 25.6x^2 - 1.28x^3$$

$$\left[\frac{102.4x^3}{3} - \frac{5.12x^4}{4} - \frac{8x^4}{4} + \frac{0.4x^5}{5} \right]_0^8 +$$

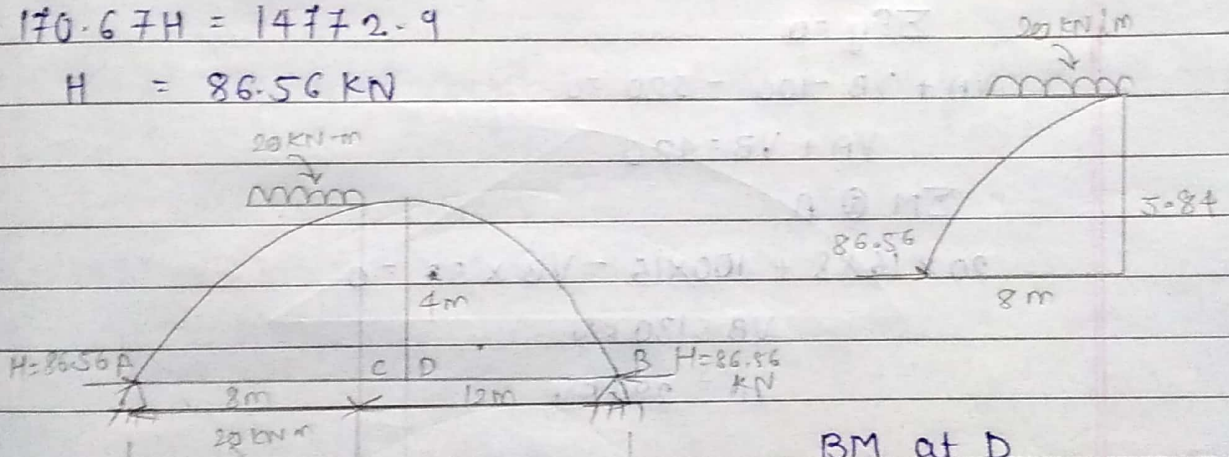
$$\left[\frac{25.6x^3}{3} - \frac{1.28x^4}{4} \right]_0^{12}$$

$$\frac{102.4 \times 8^3}{3} - \frac{5.12 \times 8^4}{4} - \frac{8 \times 8^4}{4} + \frac{0.4 \times 8^5}{5} + \frac{25.6 \times 12^3}{3} - \frac{1.28 \times 12^4}{4}$$

$$17476.26 - 5242.88 - 8192 + 2621.44 + 14745.6 - 6635.52$$

$$170.67H = 14772.9$$

$$H = 86.56 \text{ KN}$$



BM at D

$$= 128 \times 8 - 86.56 \times 3.96$$

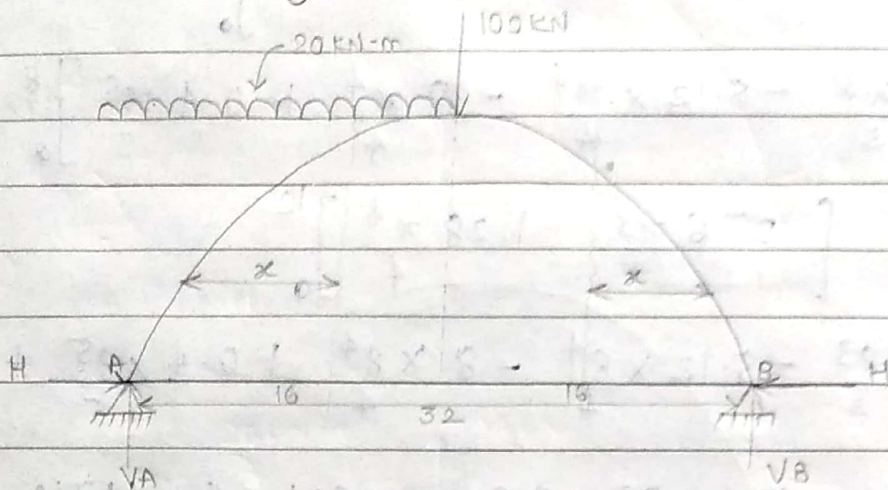
$$- 20 \times 8 \times 4$$

$$= 41.22 \text{ KN-m}$$

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W-16

Q 2 hinge parabolic arch of 32m span and the central rise of 8m as shown in fig. Calculate the horizontal thrust and draw BMD



$$\sum F_y = 0$$

$$V_A + V_B - 100 - 320 = 0$$

$$V_A + V_B = 420$$

$$\sum M @ A$$

$$20 \times 16 \times 8 + 100 \times 16 - V_B \times 32 = 0$$

$$V_B = 130 \text{ kN}$$

$$V_A = 290 \text{ kN}$$

Portion	Origin	Limit	Moment
AC	A	0-16	$290x - 20x^2/2$
BC	B	0-16	$130x$

$$y = \frac{4hx}{l^2} (l-x)$$

$$= \frac{4 \times 8 \times x}{32^2} (32-x)$$

$$= \frac{1}{32} x (32-x)$$

$$= \int_0^L My \, dx$$

$$= \int_0^{16} (290x - 10x^2) \times \frac{1}{32} x(32-x) \, dx + \int_0^{16} 130x \times \frac{1}{32} x(32-x) \, dx$$

$$= \int_0^{16} (290x - 10x^2)(x - 0.031x^2) \, dx + \int_0^{16} 130x(x - 0.031x^2) \, dx$$

$$= \int_0^{16} (290x^2 - 8.99x^3 - 10x^3 + 0.31x^4) \, dx + \int_0^{16} (130x^2 - 4.03x^3) \, dx$$

$$= \left[\frac{290x^3}{3} - \frac{8.99x^4}{4} - \frac{10x^4}{4} + \frac{0.31x^5}{5} \right]_0^{16} + \left[\frac{130x^3}{3} - \frac{4.03x^4}{4} \right]_0^{16}$$

$$= 149826.218 + 111465.81$$

$$= 261292.03$$

$$\int y^2 \, dx = \frac{8}{15} h^2 L$$

$$= \frac{8}{15} \times 8^2 \times 32$$

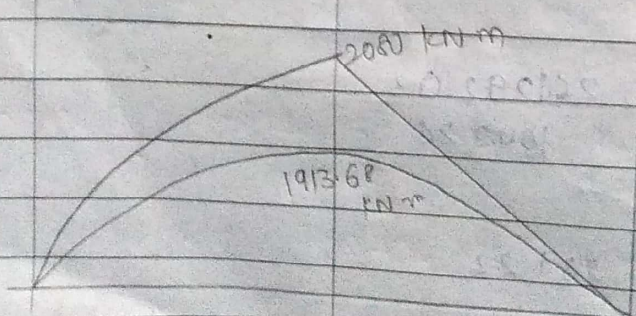
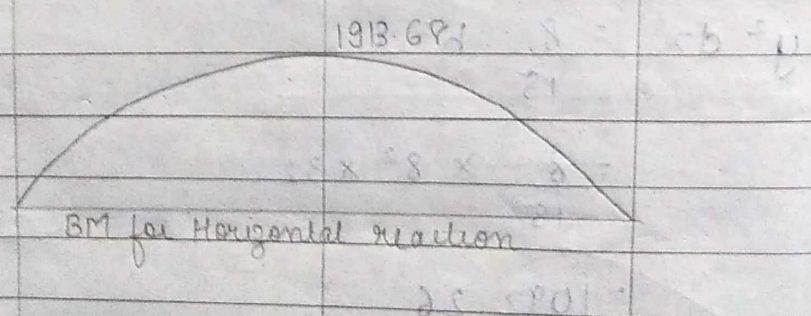
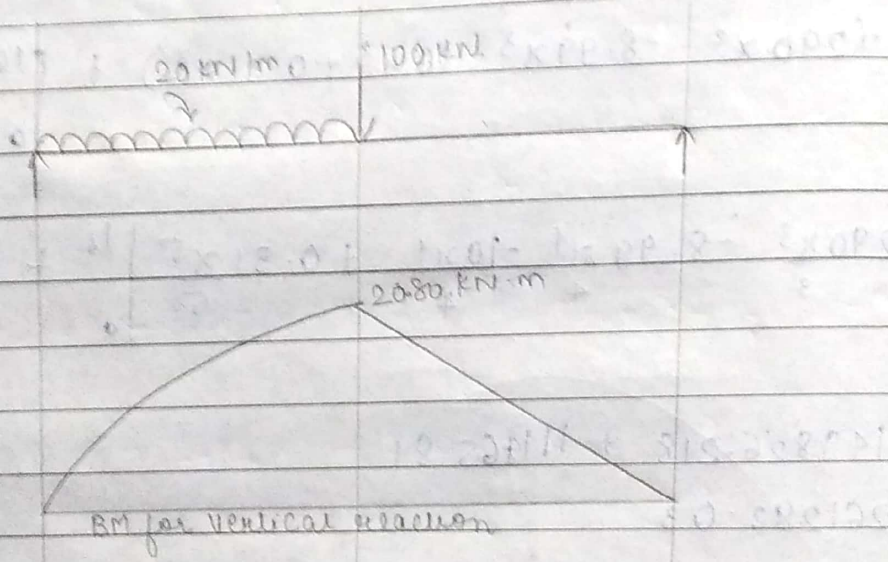
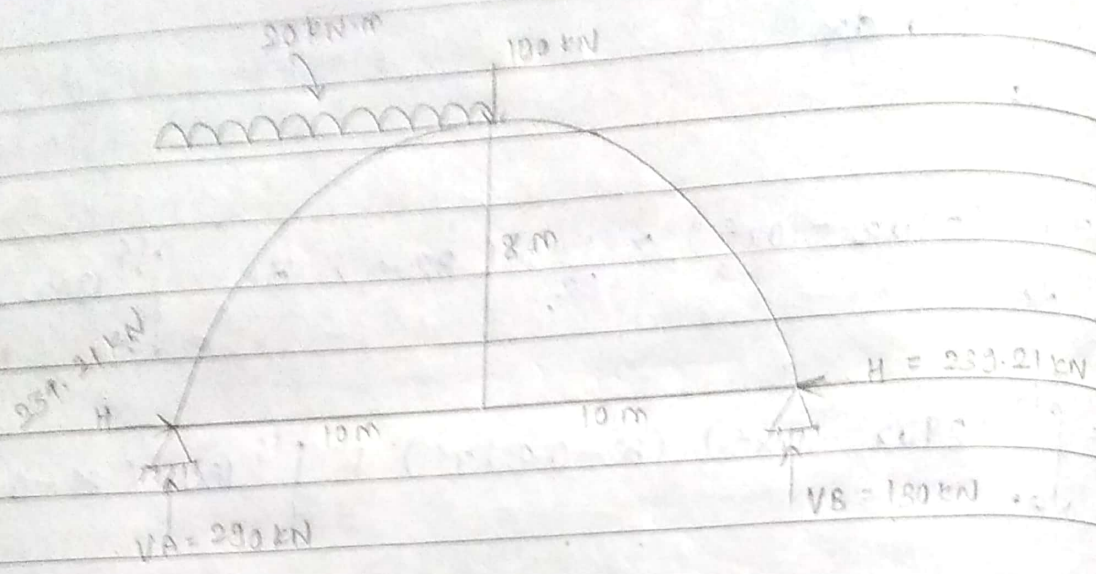
$$= 1092.26$$

$$H = \frac{261292.03}{1092.26}$$

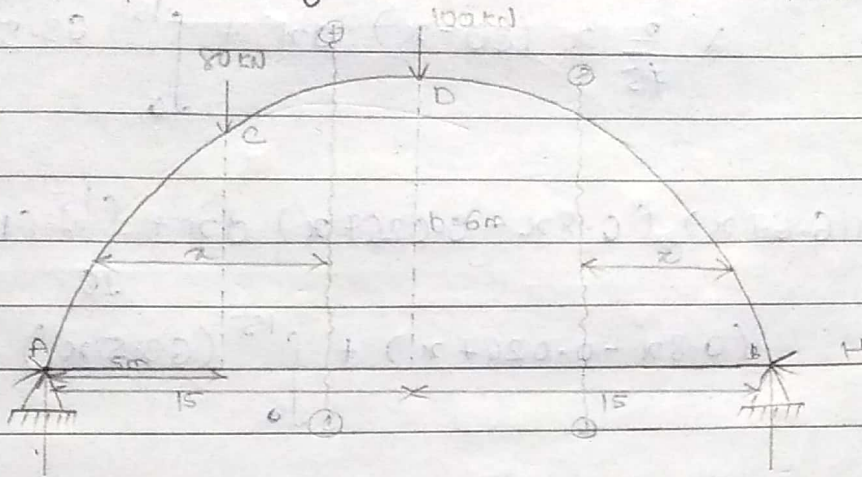
$$= 239.22$$

$$H = 239.22$$

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Q. Two hinge parabolic arch of ht 6m carries the two concentrated load of 80kN and 100kN acting at 5m and 15m from the left support calculate the horizontal thrust and BMD the total span of the arch = 30m



$$\sum f_y = 0$$

$$V_A + V_B - 80 - 100 = 0$$

$$V_A + V_B = 180$$

$$\sum M @ A = 0$$

$$(80 \times 5) + (100 \times 15) - V_B \times 30 = 0$$

$$V_B = 63.33 \text{ kN}$$

$$V_A = 116.67 \text{ kN}$$

Portion	Origin	Limit	Moment
AC	A	0-5	$116.67x$
CD	C	5-15	$116.67x - 80(x-5)$
BD	B	0-15	$63.5x$

$$y = \frac{4hx}{l^2} (l-x) = \frac{4 \times 6 \times x}{30^2} (30-x)$$

$$= \frac{2}{75} x(30-x) = 0.8x - 0.0267x^2$$

$$= \int_0^1 My \, dx$$

$$= \int_0^5 (116.67x) \times \frac{2}{75} x(30-x) \, dx + \int_5^{15} (116.67x - 80) \times \frac{2}{75} x(30-x) \, dx + \int_0^5 63.5x \times \frac{2}{75} x(30-x) \, dx$$

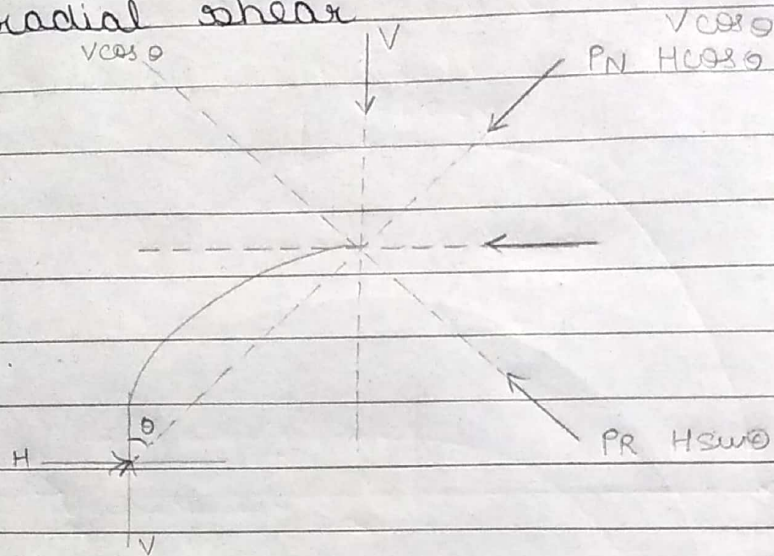
$$= \int_0^5 (116.67x) (0.8x - 0.0267x) \, dx + \int_5^{15} (116.67x - 80x - 400) (0.8x - 0.0267x) \, dx + \int_0^5 (63.5x) (0.8x - 0.0267x) \, dx$$

$$= \int_0^5 (93.336x^2 - 3.1150x^2) \, dx + \int_5^{15} (93.336x^2 - 3.1150x^2 - 64x^2 - 2.136x^2 - 320x + 10.68x) \, dx + \int_0^5 (50.8x^2 - 1.695x^2) \, dx$$

$$= \left(\frac{93.336x^3}{3} - \frac{3.1150x^3}{3} \right)_0^5 + \left(\frac{93.336x^3}{3} - \frac{3.1150x^3}{3} - \frac{64x^3}{3} - \frac{2.136x^3}{3} - \frac{320x^2}{2} + \frac{10.68x^2}{2} \right)_5^{15} + \left(\frac{50.8x^3}{3} - \frac{1.695x^3}{3} \right)_0^{15}$$

$$= \left(\frac{93.336 \times 5^3}{3} - \frac{3.1150 \times 5^3}{3} \right) + \left(\frac{93.336 \times 15^3}{3} - \frac{3.1150 \times 15^3}{3} - \frac{64 \times 15^3}{3} - \frac{2.136 \times 15^3}{3} - \frac{320 \times 15^2}{2} + \frac{10.68 \times 15^2}{2} - \frac{93.336 \times 5^3}{3} + \frac{3.1150 \times 5^3}{3} + \frac{64 \times 5^3}{3} + \frac{2.136 \times 5^3}{3} + \frac{320 \times 5^2}{2} - \frac{10.68 \times 5^2}{2} \right) + \left(\frac{50.8 \times 15^3}{3} - \frac{1.695 \times 15^3}{3} \right)$$

* Two hinge arch subjected to normal thrust and radial shear



$$P_R = H \sin \theta - V \cos \theta$$

$$P_N = H \cos \theta + V \sin \theta$$

$$y = \frac{4bx}{l^2} (l-x)$$

$$y = \frac{4bxl}{l^2} - \frac{4bx^2}{l^2}$$

$$= \frac{4b}{l} - \frac{4bx^2}{l^2}$$

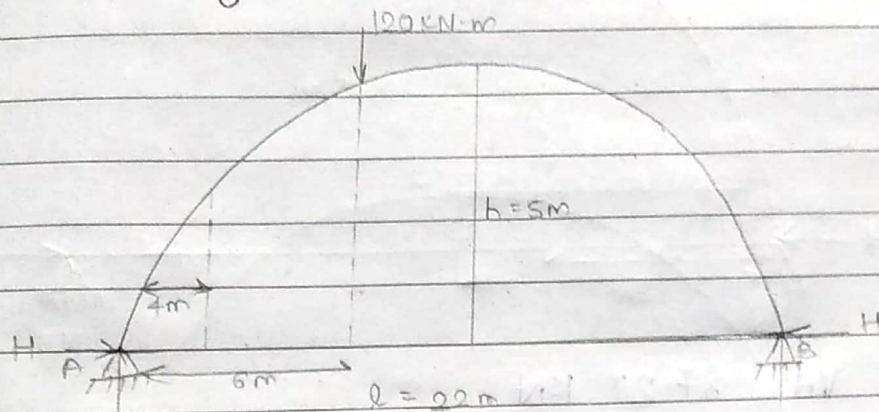
$$= \frac{4b}{l} - \frac{8bx}{l^2}$$

$$\tan \theta = \frac{dy}{dx}$$

$$\theta = \tan^{-1} \left(\frac{dy}{dx} \right)$$

* Numericals *

Q1 calculate the normal thrust and radial shear for the given two hinged area at the section 4m from the left support



$$V_A = 87.27 \text{ kN}$$

$$H = 78.58 \text{ kN}$$

$$\frac{dy}{dx} = \frac{4h}{l} = \frac{8hx}{l^2}$$

$$= \frac{4 \times 5}{22} = \frac{8 \times 5 \times 4}{(22)^2}$$

$$\frac{dy}{dx} = 0.578$$

$$\theta = \tan^{-1} \left(\frac{dy}{dx} \right)$$

$$\theta = \tan^{-1} (0.578)$$

$$\theta = 30.04$$

$$P_R = H \sin \theta - V \cos \theta$$

$$= 78.58 \sin (30.04) - 87.27 \cos (30.04)$$

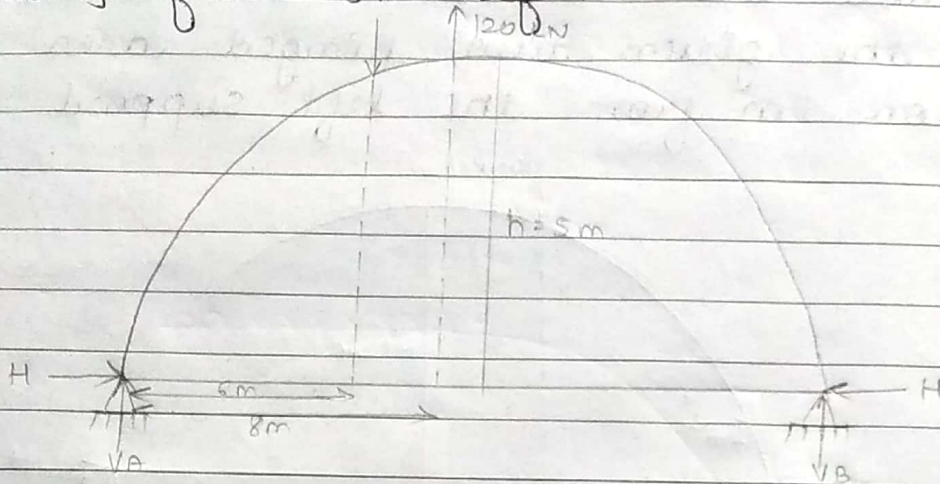
$$P_R = -36.21 \text{ kN}$$

$$P_N = H \cos \theta + V \sin \theta$$

$$= 78.58 \cos (30.04) + 87.27 \sin (30.04)$$

$$P_N = 111.71 \text{ kN}$$

Q.2 calculate the radial shear and normal reaction for the above numerical section passing from 8m left



$$V_A = 87.27 \text{ KN}$$

$$H = 78.58 \text{ KN}$$

$$V = -120 + 87.27 \text{ KN} = -32.73$$

$$\frac{dy}{dx} = \frac{4h}{L} - \frac{8hx}{L^2}$$

$$= \frac{4 \times 5}{8} - \frac{8 \times 5 \times 8}{8^2}$$

$$\frac{dy}{dx} = 0.24$$

$$\theta = \tan^{-1} \left(\frac{dy}{dx} \right)$$

$$= \tan^{-1} (0.24)$$

$$\theta = 13.49^\circ$$

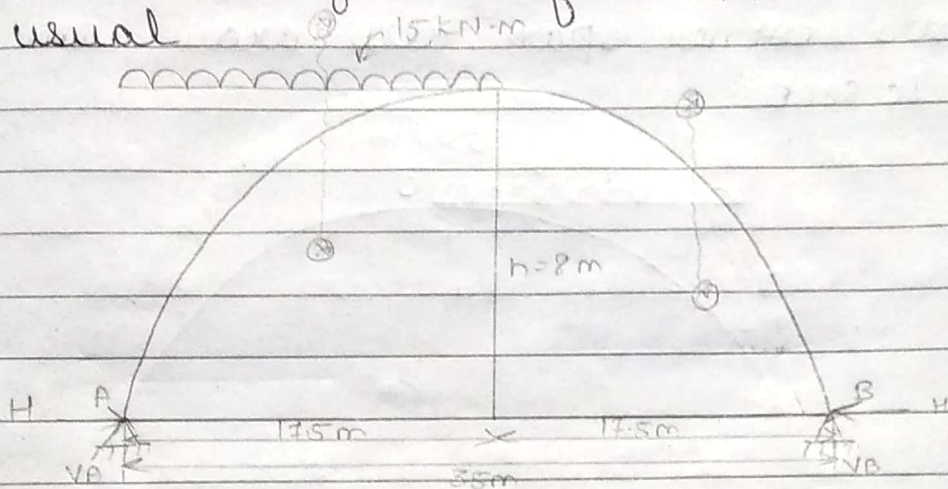
$$P_R = 78.58 \sin (13.49^\circ) - 32.73 \cos (13.49^\circ)$$

$$= -13.49 \text{ KN}$$

$$P_N = 78.58 \cos \theta + V \sin \theta$$

$$P_N = 68.77 \text{ KN}$$

Q.3 A two hinged parabolic arch with 15m span and 8m rise is subjected to UDL of 15kN/m over the left hand span. Find the reaction at support normal and radial stress at the section 12m from left support. Take $I = I_{sec}$ with usual



$$\sum F_y = 0$$

$$V_A + V_B - 262.5 = 0$$

$$V_A + V_B = 262.5$$

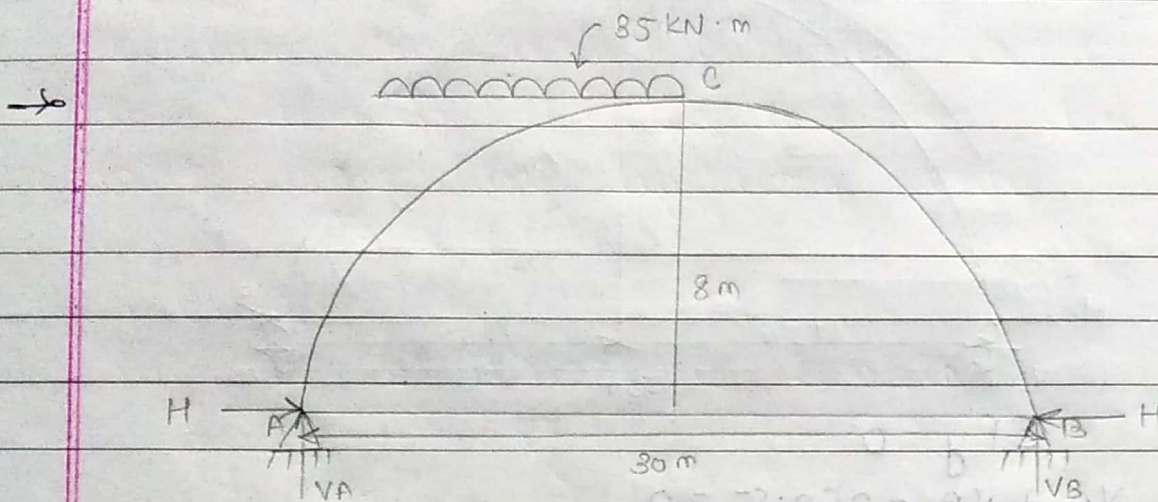
$$\sum M @ A = 0$$

$$262.5 \times 8.75 - V_B \times 15 = 0$$

$$V_B = 65.625 \text{ KN}$$

$$V_A = 196.875 \text{ KN}$$

Q. A two hinge parabolic arch of span 30m and rise 8m is loaded with the UDL of the intensity 35 kN/m over the left half span. Determine the value of horizontal thrust, radial shear and normal thrust at the left centre corner span also draw the BMD use $I = I_c \sec \theta$.



$$V_A + V_B = 525$$

$$\sum M @ A = 0$$

$$35 \times 15 \times \frac{15}{2} - V_B \times 30 = 0$$

$$V_B = 131.25 \text{ KN}$$

$$V_A = 393.75 \text{ KN}$$

$$y = \frac{4hx}{l^2} (l-x)$$

$$= \frac{4 \times 8 \times x}{30^2} (30-x)$$

$$= 1.067x - 0.035x^2$$

$$y^2 = \frac{8}{15} h^2 l$$

$$= \frac{8}{15} \times 8^2 \times 30 = 1024$$

Portion	Origin	Limit	moment
AC	A	0-15	$393.75x - \frac{35x^2}{2}$
BC	B	0-15	$131.25x$

$$\int My \, dx$$

$$\int_0^{15} (393.75x - 17.5x^2)(1.067x - 0.035x^2) \, dx + \int_0^{15} 131.25x(1.067x - 0.035x^2) \, dx$$

$$\int_0^{15} 420.13x^2 - 13.78x^3 - 18.67x^3 + 0.6125x^4 \, dx + \int_0^{15} 140.04x^2 - 4.59x^3 \, dx$$

$$\left[\frac{420.13x^3}{3} - \frac{13.78x^4}{4} - \frac{18.67x^4}{4} + \frac{0.6125x^5}{5} \right]_0^{15} +$$

$$\left[\frac{140.04x^3}{3} - \frac{4.59x^4}{4} \right]_0^{15}$$

$$\frac{420.13 \times 15^3}{3} - \frac{13.78 \times 15^4}{4} - \frac{18.67 \times 15^4}{4} + \frac{0.6125 \times 15^5}{5}$$

$$+ \frac{140.04 \times 15^3}{3} - \frac{4.59 \times 15^4}{4}$$

$$472646.25 - 174403.125 - 236292.188 + 93023.438$$

$$+ 157545 - 58092.188$$

$$\frac{254427.187}{1024}$$

$$H = 248.46$$

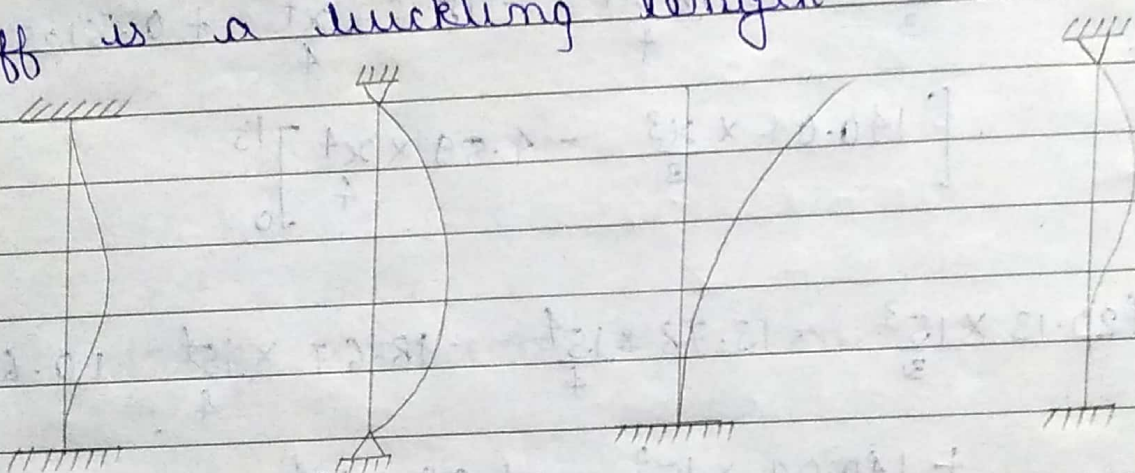
* The load carrying capacity of the column can be estimate by using two theory

1 Euler's Theory

Euler theory is used to determine the load carrying capacity of column which is derived by considering the column as long column and as per this theory the failure of the column takes place by buckling only

$$P = \alpha^2 EI / l_{eff}^2$$

l_{eff} is a buckling length



$$l_{eff} = l/2$$

$$P = \alpha^2 EI / (l/2)^2$$

$$l_{eff} = l$$

$$P = \alpha^2 EI / l^2$$

$$l_{eff} = 2l$$

$$P = \alpha^2 EI / (2l)^2$$

$$l_{eff} = L / \sqrt{2}$$

$$P = \alpha^2 EI / (l/\sqrt{2})^2$$

2 Rankine's Theory

It is the modification of Euler's theory by taking into account the crushing failure along with buckling. Euler's theory only buckling failure is considered

which is valid for long column only while Rankine's theory both crushing and buckling failures are considered which are applicable for any type of column (long and short column)

$$\frac{1}{P_R} = \frac{1}{P_c} + \frac{1}{P_E}$$

P_R = Rankine's load

P_c = crushing load = $\sigma_c \cdot A$

P_E = Euler's load

$$P_E = \frac{\pi^2 EI}{l_{eff}^2}$$

$$\frac{1}{P_R} = \frac{1}{P_E + P_c}$$

$$= \frac{P_c P_E}{P_E + P_c}$$

$$= \frac{P_c}{P_E + P_c}$$

$$= \frac{P_c}{P_E + P_c}$$

$$= \frac{P_c}{P_E + P_c}$$

$$= \frac{P_c}{P_E + P_c}$$

$$P_R = \frac{P_c}{1 + \frac{P_c}{P_E}}$$

$$= \frac{P_c}{1 + \frac{P_c}{P_E}}$$

$$= \frac{P_c}{1 + \frac{P_c}{P_E}}$$

$$= \frac{\sigma_c \cdot A}{1 + \frac{\sigma_c \cdot A}{P_E}}$$

$$= \frac{\sigma_c \cdot A}{1 + \frac{\sigma_c \cdot A}{P_E}}$$

$$= \frac{\sigma_c \cdot A}{1 + \frac{\sigma_c \cdot A}{P_E}}$$

$$= \frac{\sigma_c \cdot A}{1 + \frac{\sigma_c \cdot A}{P_E}}$$

$$= \frac{\sigma_c \cdot A}{1 + \frac{\sigma_c \cdot A}{P_E}}$$

$$= \frac{\sigma_c \cdot A}{1 + \frac{\sigma_c \cdot A}{P_E}}$$

$$= \frac{\sigma_c \cdot A}{1 + \frac{\sigma_c \cdot A}{\frac{\pi^2 EI}{l_{eff}^2}}}$$

$$= \frac{\sigma_c \cdot A}{1 + \frac{\sigma_c A}{\alpha^2 EI} \times l_{eff}^2}$$

$$= \frac{\sigma_c \cdot A}{1 + \frac{\sigma_c}{\alpha^2 E} \left(\frac{l_{eff}}{I/A} \right)^2}$$

$$= \frac{\sigma_c \cdot A}{1 + \frac{\sigma_c}{\alpha^2 E} \left(\frac{l_{eff}}{\alpha_{min}} \right)^2} \dots \dots M \text{ and } D^{\wedge} \text{ by } A$$

$$P_R = \frac{\sigma_c \cdot A}{1 + \alpha \left(\frac{l_{eff}}{\alpha_{min}} \right)^2}$$

Take $\frac{\sigma_c}{\alpha^2 E} = \alpha = \text{constant}$

$$\therefore \sqrt{\frac{I}{A}} = \alpha_{min}$$

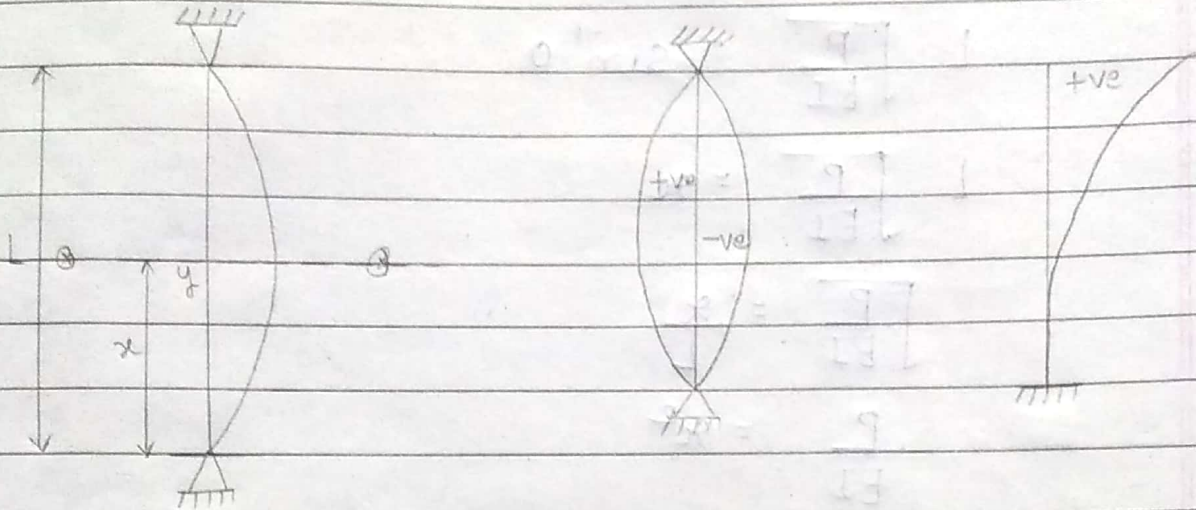
* The basic equation of Bending

$$\frac{M}{I} = \frac{\sigma}{y}$$

$$M = EI \frac{d^2 y}{dx^2}$$

Derive the eqⁿ for the effective length of column having both the end hinged
OR Derive the eqⁿ for load carrying capacity of the column having both end hinged

Consider a column AB of length L which is hinged at A and B subjected to the axial load P as shown in fig. Due to the application of load column will buckle as shown.



Consider a section X-X at the distance x from point A where y represents the deflection in column.

$$M = -Py$$

$$EI \frac{d^2y}{dx^2} + Py = 0$$

$$\frac{d^2y}{dx^2} + \frac{Py}{EI} = 0$$

The general solution of the above differential eqⁿ will be written as

$$y = c_1 \cos \left(x \sqrt{\frac{P}{EI}} \right) + c_2 \sin \left(x \sqrt{\frac{P}{EI}} \right)$$

Boundary conditions

1) At $x=0$, $y=0$

$$c_1 = 0$$

$$2) \text{ At } x=1 \quad y=0$$

$$C_2 \sin \left(\sqrt{\frac{P}{EI}} \right) = 0$$

C_2 can not be equal to zero

$$\sin \left[L \left(\sqrt{\frac{P}{EI}} \right) \right] = 0$$

$$L \sqrt{\frac{P}{EI}} = \sin^{-1} 0$$

$$L \sqrt{\frac{P}{EI}} = x$$

$$\sqrt{\frac{P}{EI}} = \frac{x}{L}$$

$$\frac{P}{EI} = x^2$$

$$P = \frac{x^2 EI}{L^2}$$

$$= \frac{x^2 EI}{l_{eff}^2}$$

comparing the basic eqⁿ of eulers then
 $l_{eff} = l$